

# A perturbation theory for the shape of central force orbits

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**Abstract.** The two body central force orbit can be solved exactly only for the gravitational and simple harmonic oscillator potentials. When one discusses nonlinear oscillators, the trajectory in space-time can be found by various kinds of perturbative techniques- one of the most prominent ones being the Lindstedt-Poincare perturbation theory. In this work we show that a Lindstedt-Poincare like perturbation theory can be set up for the shape of a general central force orbit by working round a circular orbit. One also gets an answer for spatial frequency by this process. The effectiveness of our technique is checked against numerical simulations.

## Introduction

We consider the dynamics of a particle of mass ‘ $m$ ’ moving in a central force field where the force is taken to be of the form  $F = -m\lambda r^{-n}$ , where  $n$  is any number such that bound orbits exist. The distance of the particle from the center of force is ‘ $r$ ’ and ‘ $\lambda$ ’ is the interaction strength. The conservation of the angular momentum (magnitude ‘ $l$ ’ per unit mass) restricts the particle to a plane. In terms of the polar co-ordinates ‘ $r$ ’ and ‘ $\theta$ ’, we have ( $u = 1/r$ )

$$\frac{d^2u}{d\theta^2} + u = \frac{\lambda}{l^2}u^{n-2} \quad (1)$$

Our perturbation theory is set up around the circular orbit characterized by  $u_0 = \left(\frac{\lambda}{l^2}\right)^{\frac{1}{3-n}}$ . The energy of the orbit is  $E_c = \frac{1}{2}l^2u_0^2\frac{n-3}{n-1}$ . Appropriate modifications are necessary for  $n = 1$ . The deviation  $u_1$  from the circular orbit defines the dimensionless quantity  $X = \frac{u_1}{u_0}$ . The variable  $X$  satisfies the dynamics

$$\frac{d^2X}{d\theta^2} + (3-n)X = \sum_{k=2}^{\infty} n^{-2}C_kX^k \quad (2)$$

The energy is expressed in terms of  $X$  as

$$\Delta E = E - E_c = \frac{1}{2}l^2u_0^2 \left[ \left( \frac{dX}{d\theta} \right)^2 + 2X + X^2 - \frac{2}{n-1} \{ (1+X)^{n-1} - 1 \} \right] \quad (3)$$

We have thus reduced the orbit equation formally to an anharmonic oscillator equation with coordinate  $X$  and timelike variable  $\theta$ . The order of perturbation theory is determined by how many powers of  $X$  is retained. In some ways this is another example of a traditional perturbation theory being used in an unexpected situation [1].

## Results and Discussion

The orbit upto second order in  $\epsilon$  is (initial conditions suitably chosen)

$$u = \left( \frac{\lambda}{l^2} \right)^{\frac{1}{3-n}} \left[ 1 - \epsilon^2 \frac{n-2}{4} + \epsilon \cos(\sqrt{3-n}\theta) + \epsilon^2 \frac{n-2}{12} \cos(2\sqrt{3-n}\theta) \right] \quad (4)$$

One gets a spatial frequency  $\Omega = \sqrt{3-n}$  within this order. We get corrections to this as we go to higher order. Note,  $\epsilon$  is the order of amplitude of  $X$  and hence is the perturbation parameter. Our results agree with the exact solutions for  $n = 2$  and  $n = -1$ . The comparison between our perturbation theory result and the numerically obtained trajectory, spatial frequency is shown in Figure 1.

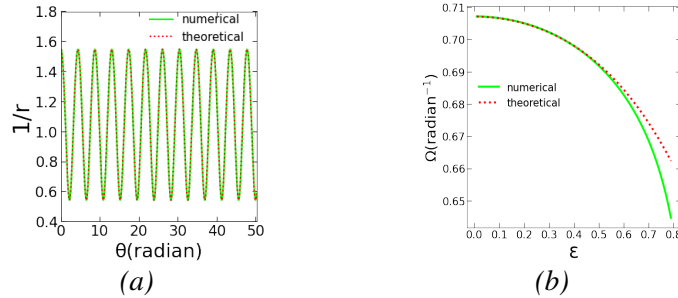


Figure 1: (a) Plot of  $\frac{1}{r}$  as a function of  $\theta$  for  $n = 1$ ,  $\epsilon = 0.5$ ,  $u_0 = 1$  (b) Plot of  $\Omega$  as a function of  $\epsilon$  for  $n = 2.5$

## References

- [1] T Shah, R.Chattopadhyay, K.Vaidya, S.Chakraborty (2015) Conservative Perturbation Theory for Non Conservative Systems. *Phys Rev E* **92** 062927